## SiC Matrix Composites for High Temperature Hypersonic Vehicle Applications, Phase I



Completed Technology Project (2010 - 2010)

#### **Project Introduction**

Durable high temperature materials are required for hypersonic engine and structural thermal protection systems. In particular, 2700°F or greater capable structural materials that can survive stresses on the order of 10 ksi (70 MPa) for at least 100 hours in an oxidizing environment have been identified as an enabling material for future hypersonic vehicles with a long term desired target of 3000°F. As these applications are structural, a strong degree of damage tolerance is desired, and thus ceramic matrix composites are the primary choice due to the desire for reduced weight, high temperature strength and oxidation resistance. Silicon carbide fiber-reinforced silicon carbide matrix (SiC/SiC) composites are believed to be the most suitable solution due to meeting the requirements with the exception of creep at the highest temperatures/loads. The proposed effort will modify the SiC fiber preform by the addition of a fraction of more creep resistant carbon fibers. The Phase I will encompass engineering the appropriate level of hybridization in a C-SiC/SiC composite, producing and evaluating the ceramic composite material for hypersonic vehicle applications, including stress rupture at temperatures of 2700°F or greater.

#### **Primary U.S. Work Locations and Key Partners**





SiC Matrix Composites for High Temperature Hypersonic Vehicle Applications, Phase I

#### **Table of Contents**

Project Introduction	1
Primary U.S. Work Locations	
and Key Partners	1
Project Transitions	2
Organizational Responsibility	2
Project Management	2
Technology Maturity (TRL)	3
Technology Areas	3
Target Destinations	3



#### Small Business Innovation Research/Small Business Tech Transfer

## SiC Matrix Composites for High Temperature Hypersonic Vehicle Applications, Phase I



Completed Technology Project (2010 - 2010)

Organizations Performing Work	Role	Туре	Location
Rolls-Royce High Temperature Composites Inc	Lead Organization	Industry	Huntington Beach, California
Langley Research Center(LaRC)	Supporting Organization	NASA Center	Hampton, Virginia

Primary U.S. Work Locations	
California	Virginia

#### **Project Transitions**

January 2010: Project Start

July 2010: Closed out

#### **Closeout Documentation:**

• Final Summary Chart(https://techport.nasa.gov/file/139930)

# Organizational Responsibility

### Responsible Mission Directorate:

Space Technology Mission Directorate (STMD)

#### **Lead Organization:**

Rolls-Royce High Temperature Composites Inc

#### **Responsible Program:**

Small Business Innovation Research/Small Business Tech Transfer

### **Project Management**

#### **Program Director:**

Jason L Kessler

#### **Program Manager:**

Carlos Torrez

#### **Principal Investigator:**

Robert Shinavski

#### **Co-Investigator:**

Robert Shinavski

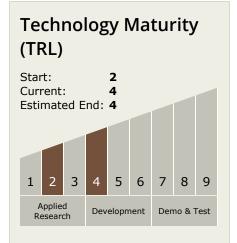


Small Business Innovation Research/Small Business Tech Transfer

## SiC Matrix Composites for High Temperature Hypersonic Vehicle Applications, Phase I



Completed Technology Project (2010 - 2010)



### **Technology Areas**

#### **Primary:**

- TX03 Aerospace Power and Energy Storage
  - ─ TX03.3 PowerManagement andDistribution
    - ☐ TX03.3.4 Advanced Electronic Parts

### **Target Destinations**

The Moon, Mars, Outside the Solar System, The Sun, Earth, Others Inside the Solar System

